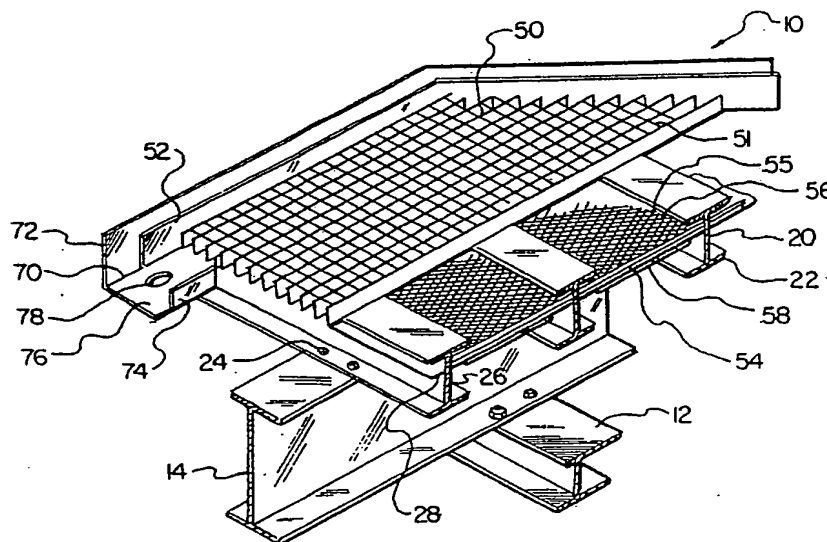




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(54) Title: FIRE RETARDANT HELICOPTER DECK



(57) Abstract

The present helicopter deck makes use of a passive fire-fighting system. Extruded aluminum beams (20) on a base (12, 14) support an upper platform (50) for supporting the helicopter. The support beams provide a space between the base and the upper platform. The upper platform is a grating (50) which permits fuel to pass through it. The support beams support batts (54) of thin, spaced strips of high-heat conductive material (55) below the upper platform. The support beams also support deck plates (58) below and spaced from the batts. The material of the batts conducts localized heat from one location of the batts to a more even, lower temperature spread generally through the batts. The deck plates catch fuel that passes through the batts and slopes to cause the fuel to flow away from the deck.

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FIRE RETARDANT HELICOPTER DECK

BACKGROUND OF THE INVENTION

1. Field of the Invention: This invention relates to fire retardant decks for the landing and taking off of helicopters or other vertical takeoff aircraft.

5 2. Description of the Prior Art: Transportation by helicopter is becoming increasingly important to industry and government. There are large numbers of helicopter landing pads ("helipads") in use throughout the world, not only on land but also on ships and oil drilling rigs.
10 There are over 4000 helipads in the United States alone, of which somewhat less than half are simply circles painted on the ground.

Helipad designers can take many precautionary measures in constructing a helicopter landing pad for
15 maximum safety. The most hazardous portions of helicopter operation are takeoffs and landings. Even if the structural damage to a helicopter is minor in a crash landing on a pad, there is a great risk of fire because of ignition of fuel spilled from the fuel tanks, which are
20 usually located underneath the aircraft. Burning fuel flows onto the landing platform and spreads rapidly to surrounding areas. In such a situation there is an extremely serious danger of harm to personnel and further damage to the helicopter and the landing platform from
25 fire and explosion. What happens immediately after a fuel fire begins determines the ultimate course of the fire and

whether the fire may be brought under control.

Active fire-fighting systems for helicopter decks are described in the following patents.

U.S. Patent No. 4,474,130 (1984) to Birkeland
5 discloses a helicopter deck preferably for use in oil
drilling platforms. The periphery of the deck is fitted
with a gutter drained by one or more down pipes. Water
discharge orifices of a fire extinguishing system are cen-
trally located on the deck. The orifices are supplied
10 with water under pressure for flooding the deck surface,
for which control levers are arranged at the periphery of
the deck.

U.S. Patent No. 4,202,646 (1980) to Herstad discloses
a helicopter landing platform comprising a fine mesh grid
15 supported on a coarse mesh grid above a horizontal surface
such as an ordinary helicopter landing pad. A bottom
framework supports the grid structure. Conduits and
nozzles for a foam fire extinguishing agent are located in
the space between the grid work on top and the framework.
20 Burning fuel flows through the mesh and the foam ex-
tinguishes the fire. The mesh prevents the foam from
blowing away.

The main disadvantages associated with active fire-
fighting systems for helicopter landing pads are com-
25 plexity, expense, and the need for maintenance to keep
them in operational readiness. A helicopter landing deck
with an active fire-fighting system is very expensive to
construct because the materials are expensive and because
the conduits, valves, reservoirs, and other parts form a
30 complicated apparatus to set up. The system must also
have regular preventive maintenance to assure that it is
always in proper operating order.

SUMMARY OF THE INVENTION

35 In view of the limitations associated with the prior
art, it is an object of this invention to disclose and

provide a novel and improved landing deck for helicopters which incorporates a passive fire-fighting system. It is another object of the invention to provide a fire retardant helicopter deck that is significantly less complicated and expensive than the ones in conventional use. Yet another object is the provision of a fire retardant helicopter deck that does not require repeated and regular maintenance of its fire-fighting equipment. Another object of the invention is to provide a fire retardant helicopter deck that can be unmanned because the fire-fighting system is completely passive.

The present helicopter deck makes use of a passive fire-fighting system. Extruded aluminum support beams on a base support an upper platform for supporting the helicopter. The support beams are extruded aluminum with a base, a top ledge and an intermediate platform between the base and the top ledge. The support beams provide a space between the base and the upper platform. The upper platform is a grating, which permits fuel to pass through it. A connector attaches the grating to the top ledge of the support beams. The support beams also support batts of thin, spaced strips of high-heat conductive material below the upper platform. The support beams also support deck plates below and spaced from the batts. The material of the batts conducts localized heat from one location of the batts to a more even, lower temperature spread generally through the batts. The deck plates catch fuel that passes through the batts. The deck plates slope to peripheral gutters.

Any fire is extinguished because the aluminum metal-foil material conducts heat away from the fire. The batts also restrict airflow.

These and other features and advantages of the invention will be apparent to those skilled in the art from the following detailed description of a preferred embodiment.

BRIEF DESCRIPTION OF THE DRAWINGS

FIG. 1 is a perspective view of a cutaway portion of the fire retardant helicopter deck of the present invention.

FIG. 2 is a plan view, partially cutaway, of the fire retardant helicopter deck of the present invention.

FIG. 3 is a cross-sectional end view of one of the support members and the parts it supports of the fire retardant helicopter deck of the present invention.

FIG. 4 is a sectional view taken through plane 4-4 of FIG. 3 and shows the connector holding the grating to the support members.

DETAILED DESCRIPTION OF A PREFERRED EMBODIMENT

The helicopter pad of the present invention comprises several distinct components. As is common, pad 10 is octagonal (FIG. 2). The pad rests on a base. In the exemplary embodiment, the base comprises I-beams 14 (only one of which is shown in FIG. 1) on and at a right angle to I-beams 12. Fasteners 16 (FIG. 1) hold the beams together. The bottom beams 12 are part of a pre-existing structure such as a building roof, an offshore drilling platform or a ship.

A plurality of support members are mounted on the base. In the exemplary embodiment, support members 20 are extruded aluminum beams, which have a shape in cross-section shown in FIG. 3. Each support beam 20 has a base 22, which is fastened to the top of cross-beam 14 by bolts 24 (FIG. 1). Central vertical web 26 extends upward from base 22. Intermediate support platforms 28 and 30 extend outward from central web 26. Platforms 28 and 30 each have an outer region, which comprises an outer sloping surface 31, 32 (FIG. 3) that leads to recessed, horizontal extension 33, 34. Slot 35, 36 extends upward and outward from each end of the platform. The sloping surface,

horizontal surface and slot extend outward and slightly below horizontal surfaces 37 and 38.

Upper horizontal ledges 40 and 42 (FIG. 3) extend outward from the top of central web 26. The underside of each ledge has slot 43, 44.

Aluminum support beams 20 are parallel and are spaced apart approximately 0.5m in the exemplary embodiment. The support beams must be sufficiently close together to support the weight of a helicopter or any other vertical take-off and landing aircraft especially during a crash. Aluminum is used because it is light weight, is relatively strong per unit mass and can be extruded. Other appropriate alloys and materials can be substituted for the aluminum.

Connector member 60 (FIGS. 3 and 4) attaches an upper platform 50 to the top of support beams 20. The upper platform is an aluminum grating of the type shown in FIG. 1 in the exemplary embodiment, which permits liquids such as fuel to pass through it. Connector member 60 has a bottom, anchor-shaped portion 62 with two arms 63 and 64. Arm 64 is shaped to correspond to slot 43 in ledge 40 (FIG. 3), and arm 63 could fit into slot 44 in the other ledge 42 (not shown). FIG. 3 only shows one connector 60, but the exemplary embodiment uses connectors spaced along support member 20 and staggered between the two ledges 40 and 42. Knob 66 abuts outer surface 45 of ledge 40.

The upper portion 68 of connector member 60 has two vertical arms 69 and 70. Walls 51 of grating 50 rest on and extend up from ledges 40 and 42 (FIGS. 3 and 4). Vertical arms 69 and 70 hold bolt 72 to connector 60. The top 74 of the bolt extends through clip 76. Nut 78 holds the clip in place. The clip extends over the top 53 of two adjacent grating walls 51 (FIG. 4). The numerous connectors spaced about the tops of support members 20 secure enough locations of the grating to the rest of the deck structure.

A kick plate 52 (FIG. 1) connects to the tops of support beams 20 around the outer edge of the grating 50 to indicate the outer edge of the deck.

5 Filler means are supported between base 14 and upper platform 50. The filler means in the preferred embodiment is a material sold under the trademark Explofoil. Explofoil is made of very thin foil aluminum alloy 55 that is slit and expanded to form webs 56 of hexagonally shaped openings, then layered to form an open-celled batt 54
10 (FIGS. 1 and 3). The expanded aluminum foil batts 54 are 30 mm thick, 60 mm wide, and 500 mm long in the exemplary embodiment.

Basin means in the form of deck plates 58 are mounted below the filler means. Mounting means on the support
15 members 20 support batts 54 of the filler means and deck plates 58. As FIG. 3 shows, batts 54 rest on surfaces 37 and 38 of platforms 28 and 30. Deck plates 58 extend outward horizontal extensions 33 and 34. The deck plates may have a tongue to engage grooves 35 and 36, or the deck
20 plates may be welded in place at weld 39.

Deck plates 58 should slope to the outer edge of the deck. Support beams 20 are somewhat flexible over their long length. Appropriate shims (not shown) may provide the proper slope. A slope ratio of 1:300 should be
25 sufficient for fuel drainage. The grating 50 can accommodate the small slope.

A gutter 70 is attached to support members 20 along the outer edge of the deck (FIGS. 1 and 2). Gutter 70 comprises an outer wall 72, a shorter inner wall 74 spaced
30 on base 76. One or more holes 78 in the bottom wall of the gutter allows fuel accumulating in the gutter to drain downward, away from the deck. Appropriate collectors (not shown) attach to the holes.

As FIG. 3 shows, batt 54 is spaced slightly above
35 deck plate 58 so that there is a short region 59 below batt 54 on which fuel can flow along the top of deck plate

58.

Safety net 90 surrounds the octagonally shaped deck (FIG. 2). The net is inclined upward 12.6° from the surface of grating 50. Such safety nets and corresponding means for attaching them to the deck are known in the helicopter pad art.

The helicopter deck of the present invention functions as a passive fire-fighting system in the following manner. Any fuel that spills onto grating 50 flows through it into porous batts 54 and then to the deck plates 58. Because the deck plates are sloped toward the outer edges of the deck, fuel that reaches the deck plates is drained into gutter 30 and accumulated fuel in the gutter is drained out.

Any fuel that is spilled on deck 50 flows quickly through the grating. If the fuel ignites, the rapid flow of fuel through the grating away from the helicopter or other objects on the grating minimizes the amount of fuel available for combustion on the deck. The fuel that flows down from grating 50 then reaches batt 54. At this point the fuel is still ignited. As the fuel reaches the batts, however, ignition is suppressed because the thin aluminum foil transmits the heat generated during combustion throughout the batt where it dissipates rapidly. As the material transmits the heat away from the region where fuel is burning, the fuel falls below its ignition temperature. The structure of the batt also inhibits the flow of air through the batt so that wind or convection currents cannot drive combustion. High winds are often a problem on off-shore oil platforms, where, for example in the North Sea winds exceed twenty knots (37 km/hr) 37% of the time. If fire retardant foam is used, the batting material tends to trap the foam and prevent the wind from blowing it away.

The cooled fuel then drips onto deck plate 58. Even if the fuel is still burning at this point, the batting

tends to remove the heat of combustion from the region around the burning fuel to minimize damage. The relatively closely spaced support members 20 (FIG. 1) also tend to keep the burning region localized. Fuel that spills
5 between two adjacent support members 20 can flow between those members, but it cannot flow to regions between adjacent members.

A scaled-down version of the helicopter deck of the present invention was tested. A tray-like metal frame approximately 2m x 2m x 20cm deep had three parallel pipes
10 in the bottom to serve as supports for a metal grating. Explofoil batts were laid on the bottom of the tray in the trough-like spaces between the bottom of the tray and the grating, between the support pipes. Aviation fuel was
15 poured on the top grating and ignited.

The initial flaring of the burning fuel reached a peak in intensity approximately ten seconds after ignition. The flames diminished after thirty seconds and were practically extinguished after fifty seconds. After
20 sixty seconds the fire was 90% out.

When the fire was completely extinguished, the grating was cool enough so that someone with shoes on could stand on the grating. The batts and the other structure was not damaged. Large pools of unburned fuel
25 had collected below the batts. Of course, gutter 70 of the present invention would drain the pools away. One could handle the batts seventy seconds after the fire had been started; they were only warm to the touch. The metal bottom of the test tray, corresponding to the invention's
30 deck plate, was cool. The grating was still too hot to handle with bare hands 120 seconds after the ignition of the spilled fuel.

A preferred embodiment of a novel and improved fire retardant helicopter deck which is a highly effective
35 safety installation for helicopter landings and takeoffs has thus been shown and described. Numerous modifications

and alternative embodiments will occur to those skilled in the art.

WE CLAIM:

1. In a helicopter landing deck comprising a base (12, 14), a plurality of support members (20) on the base, and an upper platform (50) for supporting the helicopter, the support members (20) providing a space between the base and the upper platform, the upper platform comprising a grating, which permits liquid to pass through it, the improvement comprising the provision of:

filler means (54) between the base and the upper platform, basin means (58) below the filler means and mounting means on the support members for supporting the filler means and the basin means in the space below the upper platform, the filler means comprising spaced-apart layers (54) of high-heat conductive material (55) for permitting liquid passing through the upper platform to pass through the filler means, the material of the filler means conducting localized heat from one location of the filler means to a more even, lower temperature spread generally through the filler means, the basin means catching liquid that passes through the filler means.

2. In the helicopter landing deck of claim 1, the improvement further comprising the provision of each support member (20) having a support base (22), an upright web (26) extending upward from the base, intermediate support platforms (28, 30) extending outward from the upright web above the base of the support member and an upper ledge member (40, 42) at the top of the upright web.

3. In the helicopter landing deck of claim 2, the improvement further comprising the provision of each intermediate support platform (28, 30) having an outer region which comprises an outer sloping surface (31, 32) and a recessed, horizontal extension (33, 34), the basin means (58) having side edges, the side edges of the basin means being supported on the horizontal extension of the

intermediate support platform.

4. In the helicopter landing deck of claim 3, the improvement further comprising the provision of the filler means (54) being mounted on the intermediate support platforms (28, 30) and extending towards the central web (26) of the support member (20) beyond the outer sloping surface (31, 32) above the basin means (58).

5. In the helicopter landing deck of claim 1, the improvement further comprising the provision of gutter means (70) around the periphery of the deck connected to the basin means (58) for receiving liquid that flows off the basin means.

6. A helicopter landing pad comprising:

- a. a grating (50);
- b. a plurality of parallel, spaced deck beams (20) underlying and attached to the grating for supporting the grating;
- c. a deck plate (58) extending between each adjacent deck beam to form a plurality of troughs between adjacent deck beams and platform means (28, 30) on the deck beams for supporting the deck plates below the grating;
- d. a plurality of fire retardant batts (54) disposed between adjacent deck beams and substantially covering the deck plates; and
- e. the platform means further comprising means (31, 32, 33, 33; 34, 35, 36, 37, 38) above the deck plates.

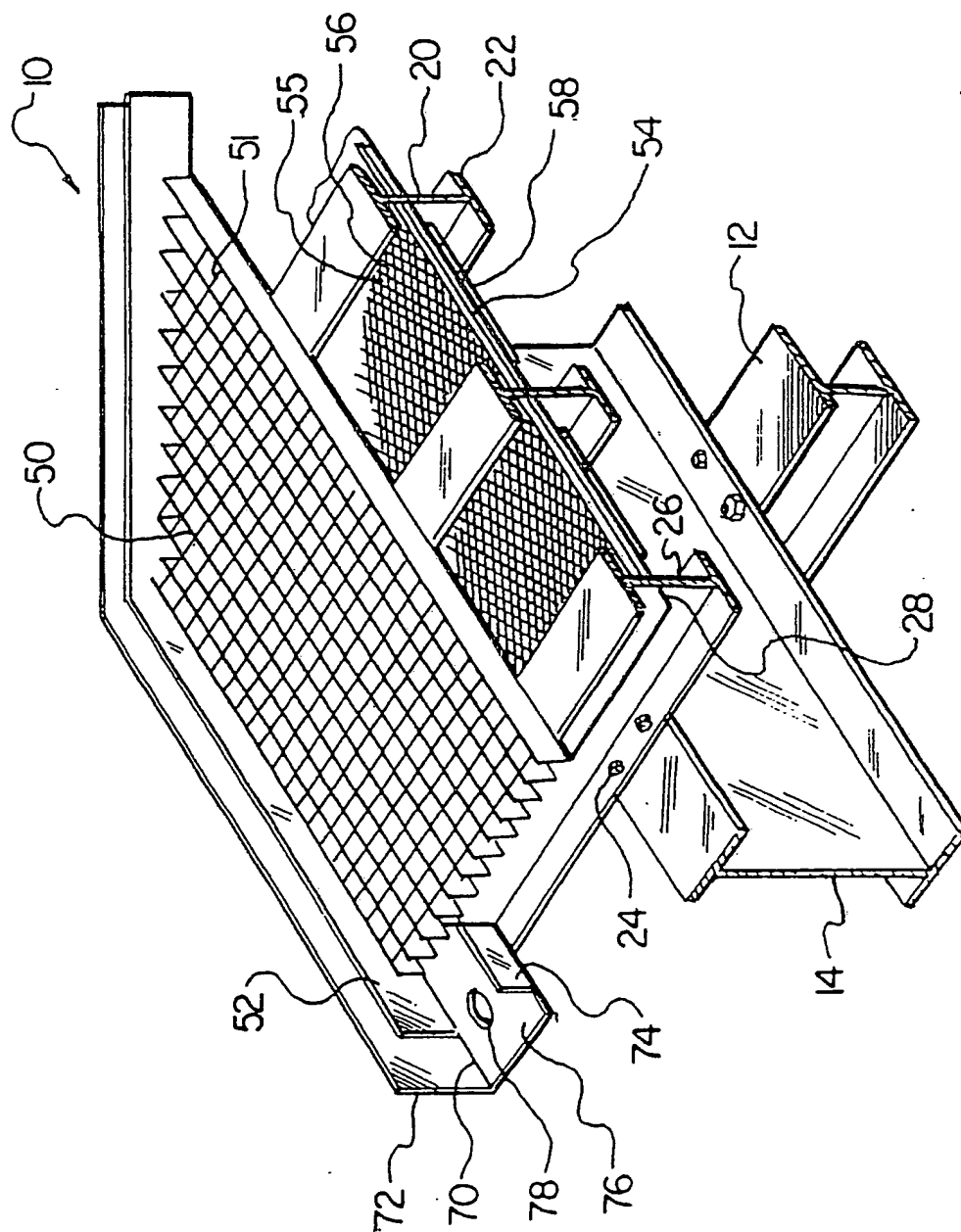
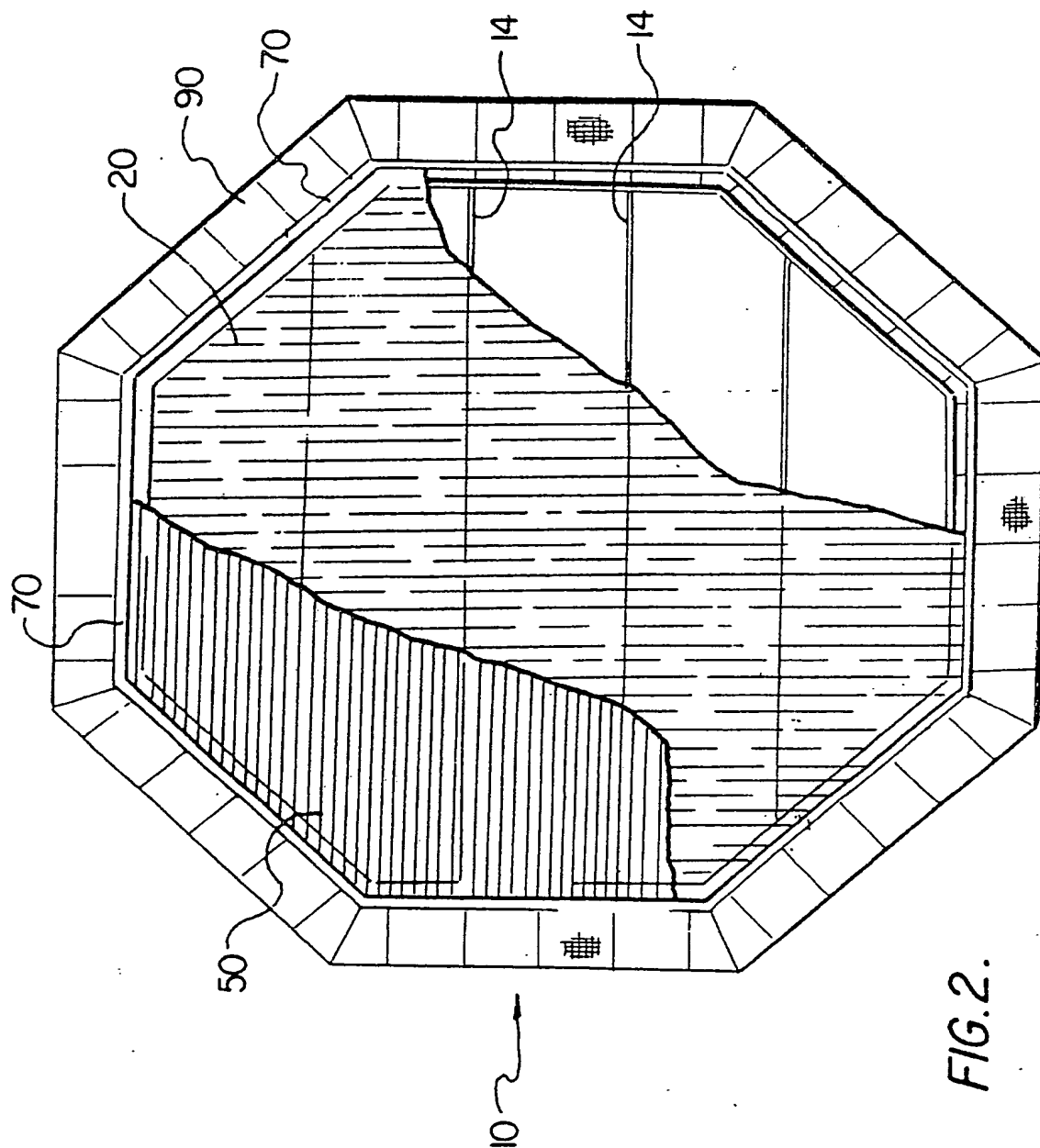
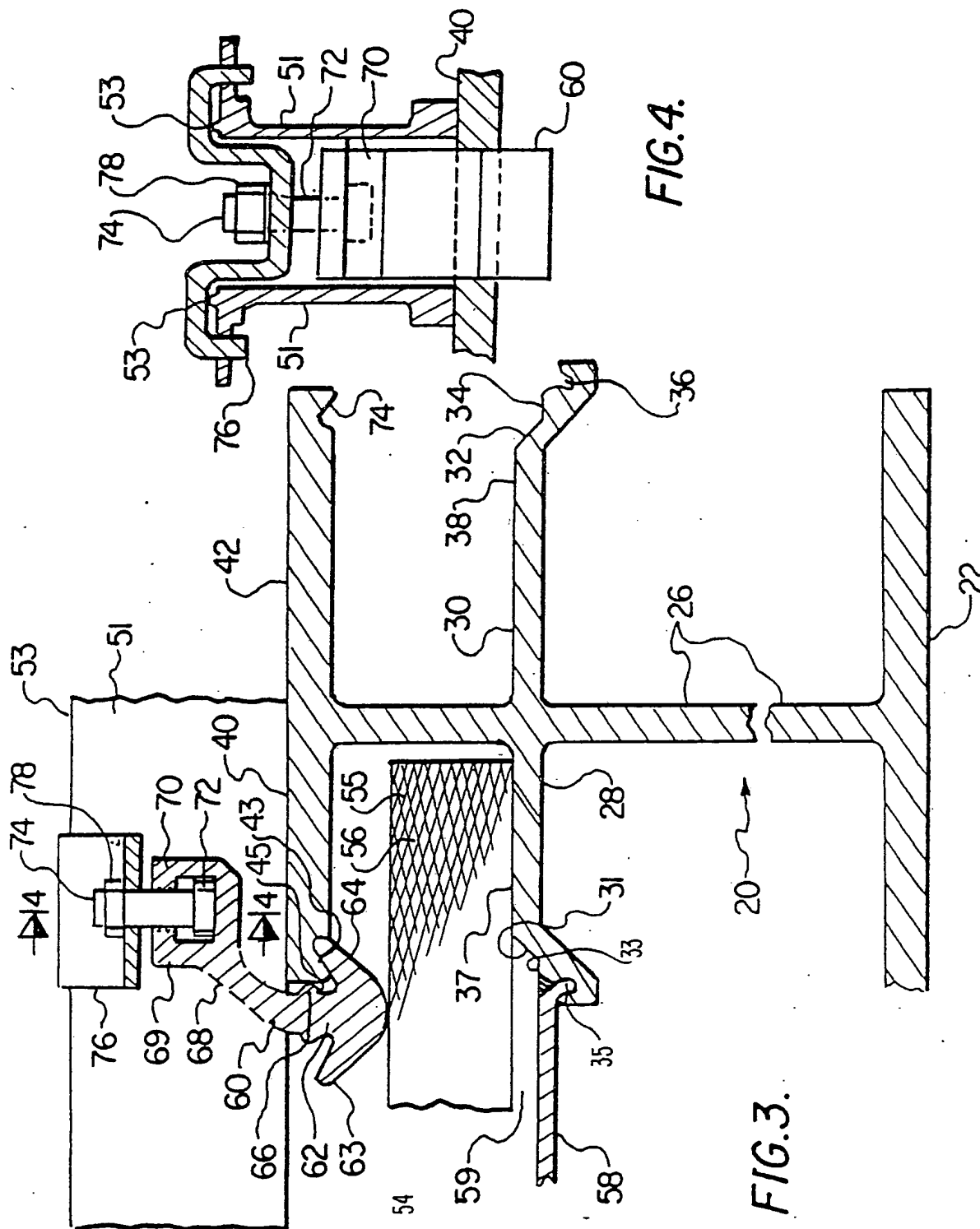



FIG. 1.





INTERNATIONAL SEARCH REPORT

International Application No PCT/US 88/00159

I. CLASSIFICATION OF SUBJECT MATTER (If several classification symbols apply, indicate all) *		
According to International Patent Classification (IPC) or to both National Classification and IPC		
IPC ⁴ : E 01 F 3/00; A 62 C 1/24; A 62 C 3/02 // E 01 C 9/00		
II. FIELDS SEARCHED		
Minimum Documentation Searched ⁷		
Classification System	Classification Symbols	
IPC ⁴	E 01 F; A 62 C; B 63 B; B 64 F; B 65 D; E 01 C	
Documentation Searched other than Minimum Documentation to the Extent that such Documents are Included in the Fields Searched *		
III. DOCUMENTS CONSIDERED TO BE RELEVANT *		
Category *	Citation of Document, ¹¹ with indication, where appropriate, of the relevant passages ¹²	Relevant to Claim No. ¹³
A	WO, A, 86/00274 (RAUFOSS AMMUNISJONSFABRIK- KER) 16 January 1986 see page 1, lines 1-4; page 2, lines 1-6, 12-15, 19-25, 28-33; page 3, lines 13-19, 23-29; page 4, lines 14-23 --	1, 2, 5, 6
A	US, A, 4202646 (HERSTAD) 13 May 1980 see column 1, lines 47-59; column 2, lines 7-24, 37-41; column 4, lines 5-12; figure cited in the application --	1, 6
A	WO, A, 82/04427 (BIRKELAND) 23 December 1982 see page 1, lines 1-3, 7-18; page 3, lines 28-30; page 4, lines 4-6, 9-11; figures 1, 2 cited in the application --	1, 5, 6
A	FR, A, 1267637 (MOREAU) 21 July 1961 see page 1, left-hand column, lines 1-22, 26, 27, 31-36 --	1, 6 ./.
<p>* Special categories of cited documents: ¹⁰</p> <p>"A" document defining the general state of the art which is not considered to be of particular relevance</p> <p>"E" earlier document but published on or after the international filing date</p> <p>"L" document which may throw doubts on priority claim(s) or which is cited to establish the publication date of another citation or other special reason (as specified)</p> <p>"O" document referring to an oral disclosure, use, exhibition or other means</p> <p>"P" document published prior to the international filing date but later than the priority date claimed</p> <p>"T" later document published after the international filing date or priority date and not in conflict with the application but cited to understand the principle or theory underlying the invention</p> <p>"X" document of particular relevance; the claimed invention cannot be considered novel or cannot be considered to involve an inventive step</p> <p>"Y" document of particular relevance; the claimed invention cannot be considered to involve an inventive step when the document is combined with one or more other such documents, such combination being obvious to a person skilled in the art.</p> <p>"A" document member of the same patent family</p>		
IV. CERTIFICATION		
Date of the Actual Completion of the International Search 20th May 1988		Date of Mailing of this International Search Report 23 JUN 1988
International Searching Authority EUROPEAN PATENT OFFICE		Signature of Authorized Officer  P.C.G. VAN DER PUTTEN

III. DOCUMENTS CONSIDERED TO BE RELEVANT (CONTINUED FROM THE SECOND SHEET)		
Category *	Citation of Document, with indication, where appropriate, of the relevant passages	Relevant to Claim No
A	FR, A, 2440892 (EXPLOSAFE) 6 June 1980 see page 1, line 19 - page 2, lines 5,25-30; page 3, lines 2-5,7-18,31 - page 4, line 11, line 34 - page 5, line 2, 15-19; page 6, lines 3-8; page 9, lines 13-20, 30-33; page 10, lines 8-11; figures 1,2 --	1,6
A	US, A, 3356256 (SZEGO) 5 December 1967 see column 1, lines 16-28, 44-47; column 2, lines 7-9,25-30,45-50,57,58; column 3, lines 24-28,30-32,35-40; column 4, lines 55-61, 64-68; figures 3-5,9 --	1,6
A	Engineering Materials and Design, vol. 19, no. 7, July 1975, IPC Industrial Press Ltd (London, GB) "Explosion risk eliminated with new expanded material", pages 28-29, see left-hand column, lines 20-30, 36-43,47-57; center-column, lines 4-7 --	1,6
A	FR, A, 2377812 (SCHMIDT-THOMEE) 18 August 1978 see page 1, lines 1-6,13-17, line 30 - page 2, lines 4,12-17,23-26, 30-35; page 3, lines 22-26,30-34, 36,37; page 4, lines 7-10; page 5, lines 7,11-16; figure 1 --	1,5,6
A	US, A, 3174411 (OESTRICH et al.) 23 March 1965 see column 1, lines 15-17,47-58; column 2, lines 5-7; column 3, lines 16-24; figure 7 -----	2,4,6

**ANNEX TO THE INTERNATIONAL SEARCH REPORT
ON INTERNATIONAL PATENT APPLICATION NO.**

US 8800159
SA 20835

This annex lists the patent family members relating to the patent documents cited in the above-mentioned international search report. The members are as contained in the European Patent Office EDP file on 10/06/88. The European Patent Office is in no way liable for these particulars which are merely given for the purpose of information.

Patent document cited in search report	Publication date	Patent family member(s)	Publication date
WO-A- 8600274	16-01-86	SE-A- 8600810	21-02-86
		DE-T- 3590306	17-07-86
		JP-T- 61502548	06-11-86
		GB-A, B 2177052	14-01-87
		SE-B- 448622	09-03-87
US-A- 4202646	13-05-80	None	
WO-A- 8204427	23-12-82	EP-A, B 0081520	22-06-83
		AU-A- 8524082	04-01-83
		US-A- 4474130	02-10-84
		AU-B- 547998	14-11-85
FR-A- 1267637		None	
FR-A- 2440892	06-06-80	US-A- 4249669	10-02-81
		DE-A- 2945067	22-05-80
		GB-A, B 2039734	20-08-80
		SE-B- 430050	17-10-83
US-A- 3356256		None	
FR-A- 2377812	18-08-78	BE-A- 863122	16-05-78
		NL-A- 7800677	25-07-78
		GB-A- 1584322	11-02-81
		CH-A- 625960	30-10-81
US-A- 3174411		None	

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